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DEVELOPMENTS IN INFRARED DETECTION AND TRACKING TECHNOLOGY AND APPLICATIONS TO MISSILES

by

He Liping





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DEVELOPMENTS IN INFRARED DETECTION AND TRACKING TECHNOLOGY AND APPLICATIONS TO MISSILES

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ABSTRACT: The article generally describes the functions, work modes, cooling methods and major performance indicators of infrared detectors. Detailed presentations are given on development of infrared detection tracking technology, in addition to several infrared detectors with development prospects in weapon systems. Moreover, applications of this technology are enumerated in the field of tactical missiles, SDI project, and in the Gulf War. Finally, several proposals are made on future developments.

Key Words: infrared detector, infrared sensor, infrared guidance, detection.

I. General Description

Infrared technology is extensively applied in industry, agriculture, medical treatment, sanitation, and everyday life; however, the principal applications of this technology are in military practice. In the fifties, development of PbS detectors was successful, used in the first-generation infrared guidance weapon system, such as Redeye, and the SA-7. In the mid-sixties, InSb detectors matured in technology, with applications of large

numbers of this detector type in weapon systems. This weapon is a second-generation infrared guidance weapon, such as Stinger missiles. After the mid-seventies, detectors of the HgCdTe infrared array and focal-plane infrared array types emerged as the third-generation infrared imaging guidance weapons. Rapid development of long-range infrared shows extensive prospects of infrared technology in missile guidance, as well as applications to detection and tracking [1].

The functions of the infrared sensor include conversion of infrared radiation energy of an object to electrical energy, or other easily detected forms of energy. In the infrared systems of guidance and detection tracking, the infrared sensor is the heart of such devices, and also the lodestar for development of infrared technology [2].

A complete infrared sensor includes an infrared detector and peripherals, such as preamplifier, cooler, optical system, rectifier cover, and electronic circuitry.

There are mainly three atmospheric windows of infrared radiation wavebands in military applications: 1 to 3, 3 to 5, and 8 to $12\mu m$, as shown in Fig. 1 [3]. At present, the last two windows are the wavebands predominantly used in devices; in particular the last window receives the emphasis in development.

Based on work modes, infrared detectors can be divided into photoconductive detectors and photovolatic detectors. After a semiconductor material absorbs incident photons, photoconductively generated characters are formed, changing the electroconductivity of the material through the photoconductive effect. Optical detectors based on this effect are called photoconductive detectors, as shown in Fig. 2. When a semiconductor material with space charge layer absorbs incident light at a certain energy, photoconductively generated

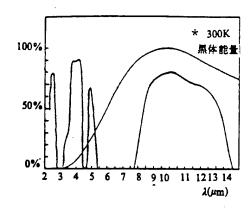


Fig. 1. Atmospheric windows sensitive
to infrared
KEY: * - black-body energy

characters are formed. Due to carrier diffusion and the function of the electric field of the PbN junctions (space charge layer), an photoconductive electric potential is formed in the the PN junction region, generating the photovoltaic effect. When this effect is used in making a detector, it is called a photovoltaic detector, as shown in Fig. 3.

The microcooler is an indispensable component of infrared technology because an infrared detector should maintain sensitivity and frequently operates at very low temperatures. For example, when an HgCdTe detector operates between 3 and 5 μ m, the cooling is 193K (-80oC); when operating between 8 and 12 μ m, cooling should be 80K (-193°C). There should be even lower temperatures for silicon-doped and platinum silicide detectors. Fig. 4 shows three cooling methods used in infrared detectors.

II. Major Performance Parameters of Infrared Detectors [3]

Major performance indicators include response rate, terminal wavelength, detection rate, and time constant, among others.

(1) Response rate: when an incident signal power is applied

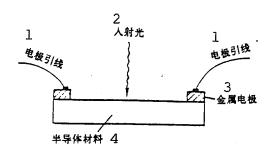


Fig. 2. Photoconductive detector
KEY: 1 - lead wire to electrode
2 - incident light 3 - metal electrode
4 - semiconductor material

normally to a sensitive surface, the mean-square-root value of the fundamental frequency voltage at the electrical output (open circuit) or the mean-square-root value of the fundamental frequency current (short circuit) with the mean-square root value of the incident radiation power is the response rate. Fig. 5 shows the ideal response rate of a detector.

- (2) Terminal wavelength: this is the radiated wavelength corresponding to 50% of the maximum response rate in the related optical spectrum response curve.
- (3) Detection rate: this is the ratio between the response rate when the detector sensitive surface is $1 cm^2$ and the noise equivalent bandwidth is 1 Hz, and the mean-square-root noise voltage or current. That is,
- (4) Time constant: this is the time period during which the photocurrent is at the maximum value after the photosensitive surface is illuminated with light. Generally, this is the time

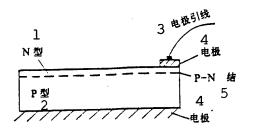


Fig. 3. Photovoltaic detector KEY: 1 - N-type 2 - P-type 3 - lead wire to electrode 4 - electrode 5 - PN junction

period when the current attains 63% of its maximum value. The smaller the parameter the better it is.

Tables 1, 2, and 3 list typical infrared detectors made in China and abroad, with the principal performance indicators for typical multielement infrared detection [4]. These data represent the principal technical level attained at present. Most detectors approach the background limit infrared property (BLIP).

Worthy of note is how to properly select and rationally use infrared detectors in the infrared system, which is the key to adequately exploit the performance of infrared devices. Since the various performance parameters of infrared detectors are interrelated, overall consideration should be made during selection.

The optical spectral detection rate, temperature properties, and the relationship between the field of view of infrared detectors are shown in Fig. 6 [5], and Figs. 7 and 8. This is an important factor when properly selecting appropriate detectors. When satisfying the conditions of performance requirements of

infrared systems, one should select, if possible, infrared

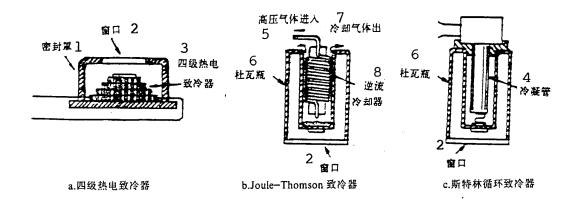


Fig. 4. Three cooling methods for infrared detectors a. four-level thermoelectric cooler b. Joule-Thomson cooler c. Stirling cycle cooler

KEY: 1 - sealed cover 2 - window 3 - four-level thermoelectric cooler 4 - cooling and condensation tube 5 - high-pressure gas inlet 6 - Dewar flask 7 - cooling gas outlet 8 - reverse-flow cooler

detectors that have high working temperature, design simplicity, high photosensitive area, high response rate, convenience in use, and low cost.

III. Progress in Infrared Detection and Tracking Technology [614]

Infrared detectors are the heart of a military infrared system. Such detectors have a unique system of high system sensitivity, good spatial resolving power, high dynamic range, high resistance to jamming, capability of 24-h round-the-clock operation in poor weather, and high capability of penetrating smoke and dust with long-wave infrared rays. Such detectors have the capability of automatic discrimination and discrimination of homing point of the target with missile-borne microcomputers [6].

Therefore, such capabilities are important for infrared imaging guidance, and detection and tracking of targets by missiles.

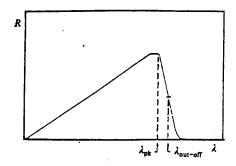


Fig. 5. Ideal response rate of detector

1. Progress in infrared detectors

In recent years, development of infrared detector technology has been fast-paced, from single-element, multielement, and array types to grow into the staring focal-plane array. Based on different scanning forms of infrared detectors, a presentation is made in the following text on the photo-device scanning system (line-array) and electronic scanning system (focal-plane array).

(1) Photo device scanning system (line-array)

Thus far, mostly the line-array scanning configuration is applied in imaging infrared (IIR) sensors. A scanning optical system is applied to scan a scene and object with part of the energy focussed sequentially on every detection element. Fig. 9 shows the development of the scanning approach.

Since the seventies, rapid development took place in linearray devices made of HgCdTe material, having met the requirements for military indicators and also commercialization. The first-generation infrared imaging guidance system, such as in the Bullpup missiles in the United States, a 16-element HgCdTe photoconductive line-array [7] was applied. By now, the shared

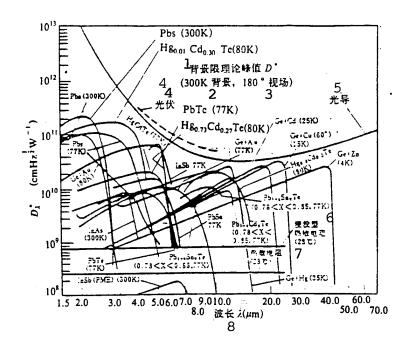


Fig. 6. Optical spectral detection rate for various kinds of infrared detectors
1 - limiting theoretical peak value of background 2 - background 3 - field of view 4 - photovoltaic 5 - photoconduction 6 - immersion-type thermistor 7 - thermistor 8 - wavelength

modules of the line-array have been developed to 60, 120, and even 180elements.

In 1974, another type of infrared device began to be developed at Murad in the U.K., called the SPRITE (Signal Processing in the Element) detector. The elements are made into long band shapes so that light spots scan along the direction of

the long bands. With an applied electric field, the photocarriers are driven to move with the scanning of the light spots. When the scanning speed of the light spots is equal to TABLE 1. Typical Performance of Foreign Infrared Detectors

	1 + 14-	ī	V :# === 68	1.00	C	探測率 D		h- 14. 41		
探測器	工作	光敏面积(cm²) 3	光谱范围 4 (μm)	峰值波长 5 ^(μm)	6%向应率 (V・W ⁻¹)	フ ^{M(数字 D} (cmHz ^{1/2} W ⁻¹)	週制频率 8 (Hz)	工作温度 9 (K)	阻抗(Ω)	上升时间
PbS	PC	6.25 × 10 ⁻⁴ ~ 1.0	0.5~3.0	2.5	8×10 ⁴	> 5 × 10 ¹⁰	750	300	10 (0.5~15)M	<u> 11 (μs)</u> < 250
PbS	PC	1 × 10 ⁻²	0.5~2.8	2.2		9×10 ⁱ⁰	1000	300	0.5M	350
PbS	PC	$0.25 \times 10^{-2} \sim 0.16$	0.7~3.5	2.6	5×10 ⁶	3×10 ¹¹	100	196	(0.5~50)M	~ 500
PbS	PC	$0.25 \times 10^{-2} \sim 0.16$	0.1~4.0	3.2	5×106	1.2×11	100	77	(1.5~50)M	300
PbSe	PC	< 0.16	1.0~5.1	4.4		7.5×10°	1000	300	2M	25
PbSe	PC		0.5~5.7	4.7	106	2.0×10 ¹⁰	1000	193	<20M	25
PbSe	PC	$0.25 \times 10^{-2} \sim 0.16$	0.8~7.5	5.5	10⁴	7×109	1000	77	~40M	60
InAs	PV	0.5	1.0~3.5	3.3	6×10³	7×10 ¹⁰	900	195	3000	1
InAs	₽V	10-2	l ~ 3.2	3.1	5 × 10 ³	4×1011	900	77	, IM	
InSb	PC	$(1 \sim 4) \times 10^{-2}$	~ 7.5	6~6.3	0.4~6	$(1 \sim 8.5) \times 10^9$	800	300	30~130	0.1
InSb	PV	10³	1.0~5.5	5.0	103	1 × 10 ¹¹	900	77	lм	0.2
inSb	PV	6×10 ⁻⁶ ~0.25	0.7~5.6	5.0	2.8(AW ⁻¹)	1.1 × 10 ¹¹	1000	~ 145	104~106	0.16
InSb	PV		~ 5.7	5.3	2(AW ⁻¹)	~ 10"	1000	77 *	i0ª	< 1.0
HgCdT	PC	10.0 × 10.0	3~5	4.6		> 1 × 10°	1000	300	~ 100	0.4
HgCdT	PC	6×10→~0.25	0.4~10		10 ² ~ 10 ³	> 2 × 10*	10.000	300~77	> 10	< 1
HgCdTe	PC	0.025 × 0.02	3~8	~5		$(1-5) \times 10^{10}$	10K	215~77	50~1K	2
HgCdTe	PC	6×10 ⁻⁵ ~0.25	0.4×20	12	10°~ 10²	7.2 × 10 ¹⁰	10K	77	10~100	< 0.5
HgCdTe	PC	5 × 10 ⁻⁵	3-12.5	11		3 × 10 ¹⁰	1000	77	50	0.25
HgCdTe	ł i	0.025 × 0.025	8-14	~10		2 × 1010	1K~10K	77	10~50	
HgCdTe		6.5×10 ⁻ⁿ -0.1		12		> 2 × 10 ¹⁰	2000	80	50	0.5
HgCdTe	1 1	6.5 × 10 ⁻⁶		12		>1.5 × 10 ¹⁰	10K	77	100	0.05
HgCdTd	1 1	0.015 × 0.025		10.6		(1~10)×10°	你死 200MHz	77	500	
HgCdTe		4×10 ⁻⁴	8~14	10.6		3 × 10 ¹⁰	12 1800	77	> 1000	10-3
HgCdTd	1	10-4	8~14			> 3 × 10 ¹⁰	1800	77	> 3000	
PbSnTe	PV	$0.008 \sim 3 \times 10^2$		10	(1.5 × 4)	$(1.5 \sim 3) \times 10^{10}$		77		1~2
					A • W-1		,	i		
PbSnTe	- 1	$(0.05 \sim 1) \times 10^{-2}$	5.5 × 12	9~11	6A • W-1	5 × 10 ¹⁰		77		0.01
PbSnTe	PV	0.8 × 10 ⁻²	8 × 13	11	3A • W⁻¹	2×10 ¹⁰		77	i	70.5
Ge:Au	PC	0.034	2~11	5.0	103	6 × 10 °	900	77	100K	0.5
Ge:Au	PC	4×10 ⁻²	1~9			6×10°	900	77	0.25M	0.1
Ge:Au	PC PC	$0.05^2 \times 0.3^2$ $10^{-4} \sim 10^{-1}$	<11	5.0		3×10°		77	(0.5~1)M	< 0.05
Ge:Hg	PC	1	2~14	11		3×10 ¹⁰	10004	. 30	10⁴	0.01~1
Ge:Hg		$0.05^2 - 0.3^2$	~14	10.6		1.7 × 10 ¹⁰		5-27	0.5M	5 × 10 ⁻³
Ge:Hg Ge:Cd	PC PC	3.14 × 10 ⁻² 10 ⁻⁴ ~ 10 ⁻¹	2~15 5~23	10~!1	10⁴	> 1.5 × 10 ¹⁰	900	25	0.2M	0.05
Ge:Cd	PC	10 ~10		22	i	3×10 ¹⁰	1000	20	104	0.01~1
Ge:Cu	PC	[5~23 -24	22		3×10 ¹⁰	10.000	20	0.534]
Si:Bi	PC	10-4-10-1	-24 4~17	18		2×10 ¹⁰	1800	5	0.5M	
Si:Ga	PC	10 - 10	4~17	15		- 10 ¹⁰	1000	20	103	İ
Si:Al	PC	10 ~ 10 10 ~ 10 -1	4~17	15		2×10 ¹⁰	1000	18	105	}
Si:Ao	PC	10 ~ 10	6~25	23	ļ	2 × 10 ¹⁰	1000	20	10'	
Si:Sb	PC	.0 10	6~33	28	İ	2×10 ¹⁰ ~10 ¹⁰	1000	4	105	
31.30			0-33		·	~ 10"	1000		10	

[KEY on next page]

KEY: 1 - detector 2 - operating mode 3 - photosensitive area 4 - optical spectral interval 5 - peak value wavelength 6 - response rate 7 - detection rate 8 - modulation frequency 9 - working temperature 10 - impedance 11 - rise time

the velocity of the carriers, the carriers generated by light at every point in an element accumulated under the influence of the electric field. Finally, the carriers accumulate at the end of the element between two electrodes, thus changing the resistive surface there with the input as the signal. This detector is also called the scanning cumulative detector. Table 4 shows its principal performance indicators.

A HgCdTe long slender band conductor with only three lead wires can obtain a performance corresponding to about 12line-array devices. Such devices considerably simplify the electronic circuitry, thus bringing benefits to the system, dimensions, costs, and reliability. At present, there are two, four, and eight bands for the SPRITE detector, corresponding to a line-array device with 50 to 100elements. However, its manufacturing technique is still under development. For the purpose of further improving the temperature and spatial resolving power of the imaging system.

(2) Electronic scanning system (focal-plane array).

Although the SPRITE detector does not have electronic circuitry, yet the detector has many advantages, since it is still of a type of mechanical scanning type. An ideal detector should have the detection elements array into a rectangular array. Acting like the human eye, the pre-lens focuses the entire scene within the detector field of view on this array. Each detection element stares at a small part of the field of

view in order to absorb as much as possible of the target infrared energy. The emergence of the charge coupled device

TABLE 2. Performance Indicators of Chinese-made Infrared Detectors

PBS PC 6×6 1.8-1.2 2-2.1 (0.8-1)×10¹ (0.5-2)×10¹ 400 空間 では、(0.1-2)3M 100-300 低子面向日 1 1 1 1 1 1 1 1 1							,					,		
PbS PC 0.4×0.7			受光而积(mm²)				再校书	探測率 D _{SOOK}			PIL(A:(Ω)		结构特	ង
PbS PC 0.4 × 0.7 1.8 × 1.2 > 2.1 (4~9)×10 ¹ (0.5~2)×10 ¹ 400 年間 (0.1~1)5M 100~200 7.万五泉 1.8 × 1.0 1.8 × 1.2 > 1.8 × 1.0 1.8 ×	乙桥▼土	75	3	³⁰⁽ 4'	5 ^(μm)	6 ^{μm})	170.00	8 (cm - 112 - W -)	Q(Hz)	(K)](11	$12^{(\mu s)}$		2111
PBS PC 6×6 1.8-3.2 2-1 (0.8-1)×10¹ (0.5-2)×10¹ 400 7計 (0.1-3)3M 100-300 旅子前房日1 7 PBS PC 0.1×0.1-6×6 0.5-3.5 2.7-2.8 10²-10² (1-3)×10² 800 300 (0.1-2)M 100-300 飛夜飛行飛 1 1 1 1 1 1 1 1 1	PbS	PC	0.4 × 0.7	T	1.8~3.2	>2.1	(4~9)×104	$(0.5 \sim 2) \times 10^9$	400	電視	(0.3~1.5)M		石英蓝板	14
PbS PC 0.1×0.1-6×6 0.5-3.3 2.1~2.4 10 ³ -10 ¹ (2~8)×10 ¹ 800 300 (0.1-2)M 100~300 品件形光 PBS PC 0.1×0.1-6×6 0.5-3.5 2.2~2.8 10 ³ -10 ¹ (1-3)×10 ³ 800 77 (0.5~20)M 1000~5000 配換算柱底面 PbS PC 0.1×0.1-6×6 0.6-4 3.1~3.4 10 ³ -10 ¹ (2~4)×10 ³ 800 77 (0.5~20)M 1000~5000 凡級費柱底面 PbS PC 1.8 1.8~3 2.5~2.7 >>×10 ³ (0.2~4)×10 ³ 800 30 40~400K 50~10 92.50c 只没是投资社股 PbS PC 1.8 1.8~3.2 2.2 (1~2)×10 ³ (2.4)×10 ³ 800 23 0.1~11M 180~250 92.50c 只投资品投资金技费 92.50c 只投资金技费 92.50c 只是投资金技费 92.50c 只是投资金技费 92.50c 只是投资金技费 92.50c 具足投资金 92.50c 具工金 92.50c 具工金	PhS	PC	6×6		1.8~3.2	>2.1			400	官湖	(0.1~3)M	100~300	精平面實口	15
PbS PC 0.1×0.1-6×6 0.5-3.5 2.7-2.8 10-10 (1-3)×10 800 17 (0.5-20)M 1000-5000 保険性孔紙 PbS PC 0.1×0.1-6×6 0.6~4 3.1-3.4 10-10 (1-3)×10 800 77 (0.5-20)M 1000-5000 保険性孔紙 PbS PC 1×1 1.8-3 2.5-2.7 >5×10 (1-2)×10 800 293 (0.1-1)M 180-250 0.2-6¢2 2.6 2.5 ×10 (1-2)×10 800 293 (0.1-1)M 180-250 0.2-6¢2 2.6 ¢26 2.2 ± (1-2)×10 (2-4)×10 800 293 (0.1-1)M 180-250 0.2-6¢2 2.2 ± (1-2)×10 (2-4)×10 800 293 (0.1-1)M 100-400 ほどはいけい 2.2 ± (1-2)×10 (2-4)×10 (2-4)×10 800 293 (0.1-1)M 100-400 ほどはいけい 2.2 ± (1-2)×10 (2-4)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 (2-4)×10 400 2.2 ± (1-2)×10 400 2.2	PbS	PC	0.1×0.1~6×6		0.5~3	2.1~2.4		(2 8) × 10	800	300	(0.1~2)M	100~300	晶体管外壳	16
PBS PC 1.7×1.7	PbS	PC	0.1 × 0.1 ~ 6 × 6		0.5~3.5	2.7 - 2.8	10'~10'	$(1-3) \times 10^{9}$	800	193	(0.5~20)M	1000~5000	配玻璃柱瓦瓶	10 17
PbS PC	PbS	PC	0.1×0.1~6×6		0.6~4	3.1~3.4	101~10	$(2 \sim 4) \times 10^{9}$	800	77	(0.5~20)M	1000 ~ 5000	配玻璃柱瓦瓶	18 17
PbS PC x 1 1-3 2.6	PbS	PC	1.7 × 1.7		1-3	2.3	60 ~ 200	$(3 \sim 9) \times 10^4$	800	300	40~400K	50~150	1	
PbS PC 1×1	PbS	PC	1 × 1		1.8~3	2.5 - 2.7	> 5 × 10	$(1 \sim 2) \times 10^{9}$	800	293	(0.1~1)M	180~250	Φ2.5Ge 浸没透镜	18
PbS PC 1×1	PbS	PC	l×1		1~3	2.6	> 7 × 10	(2-4) × 10°	800	233	(0.2~1)M	200~400	硅双球曲透镜 中6	
PbS PC 1.6×1.6 1.8×3.2 2.1 (2×3.5)×10¹ (5×8)×10² 400 24% (0.1×15)M 100×300 情況改通後 y = 3.81 1.8×3.2 2.1 (4×7)×10² (4×10)×10² 400 300 (0.3×1)M 100×300 情況改通後 y = 3.81 1.8×3.2 2.1 (4×7)×10² (4×10)×10² 800 300 (0.1×3)M 100×300 情況改通後 y = 3.81 1.8×3.2 2.1 (4×7)×10² (2×3)×10² 800 29% 200×500 <318 情況改通後 y = 5.012 1.8×3.2 2.4 2×10² (0.5×1)×10² 800 29% 200×500 <318 情況改通後 y = 5.012 1.8×3.2 2.4 2×10² (0.5×1)×10² 800 29% 200×500 <318 情況改通後 y = 5.012 1.8×3.2 2.4 2×10² (0.5×1)×10² 800 29% 200×500 <318 情況改通後 y = 5.012 1.8×3.2 2.4 2×10² (0.5×1)×10² 800 233 (0.1×1)M <16 貞双は前後 66. 四级]			29		ĺ	温差电致冷	20
PbS PC 1.6×1.6 1.8→3.2 >2.1 (2→3.5)×10¹ (5→8)×10³ 400 29年級 (0.1~1.5)M 100~300 精浸改透镜 y=3.8 PbS PC 1.6×1.6 1.8→3.2 >2.1 (4→7)×10¹ (4→10)×10³ 400 300 (0.3→1)M 100~300 情浸改透镜 y=5.012 PbS PC (1→10)×10¹ 1.6→3 2.5→2.8 10¹ 220→500 <318 背投改透镜 y=5.012 PbS PC 1×1 1.6→3.2 2.4 >1×10¹ (2→3)×10³ 800 29年間 220→500 <318 背投改透镜 p=5.012 PbS PC 1×1 1.6→3.2 2.4 >1×10¹ (2→3)×10³ 800 29年間 220→500 <318 背投改透镜 p=5.012 PbS PC 1×1 1.6→3.2 2.4 >1×10¹ (2→3)×10³ 800 29年間 220→500 <318 背投改透镜 p=5.012 PbS PC 1×1 1.5→2 4.8 >8×10¹ (2→4)×10¹ 800 233 (0.1→1)M <16 非双身间镜 中6、二级 差电制冷 24	PbS	PC	l×1		1.8~3.2	>2.2	$(1 \sim 2) \times 10^4$	$(2 \sim 4) \times 10^{9}$	800	空間	150~500K	100~300	绪投投透镜 y=1.	95min 21
PhS PC 1.6×1.6 1.8~3.2 2.1 (4~7)×10 ¹ (4~10)×10 ² 400 300 (0.3~1)M 100~300 解設改造後 y=5.012 PhS PC 1×1 1.6~3.2 2.4 >1×10 ⁴ 2.2×10 ² 800 300 (0.1~3)M	PbS	PC	1.6×1.6		1.8~3.2	>2.1	$(2 \sim 3.5) \times 10^{4}$	$(5 \sim 8) \times 10^{9}$	400 2	9宝温	(0.1~1.5)M	100~300		
PbS PC 1×1					1.8~3.2	>2.1		(4~10)×10°	400	300	(0.3~1)M	100~300		
PbSe PC I×1					1.6~3	2.5~2.8	101	> 2 × 10"	800	300	(0.1~3)M		锗浸没透镜	21
PbSe PC 1×1	PbS		1		1.6~3.2	2.4	> 1 × 10 ¹	$(2 \sim 3) \times 10^9$	800 2	9名温	200 ~ 500	< 318	带浸没透镜	22
PbSe PC 1×1	PbSe	PC	I×I		I~4.8	4.4	> 2 × 10 ¹	$(0.5 \sim 1) \times 10^4$	800	233	(0.1~1)M	<16	硅双球而镜 46	
HgCdTe PC 0.25 × 0.25 180 3 ~ 5 4.2 >3 × 10													差电制冷 24	
HgCdTe PC 0.25 × 0.25 180 3~5 4.2 >3 × 10 ¹ 25 × 10 ¹ 1000 300 100~300 ~1 競換液件 26 HgCdTe PC 0.25 × 0.25 180 3~5 4.6 >10 ² (0.5~5)× 10 ³ 1000 240 300~800 ~1 近級液件 26 300~800 ~1 近級水件 26 300~800 ~1 近級 26 300~800 ~1 近級 26 300~800 ~1 近級 26 300~800 ~1 近級 26 300~800 ~1 近級 26 300~800 ~1 近級 26 300~800 ~1 近郊 26 300~800 ~1 500	PbSe	PC	1×1		1~5.2	4.8	> 8 × 10	(2-4) × 10 ⁴	800	193	(0.2~5)M	< 30		4级温 23
HgCdTe PC 0.25 × 0.25 180 3~5 4.2 >3 × 10 ¹ 25 × 10 ¹ 1000 300 100~300 ~1 情報政条件 26 HgCdTe PC 0.25 × 0.25 180 3~5 4.6 >10 ² (0.5~5) × 10 ³ 1000 240 300~800 ~1 近級条件 26 300~800 ~1 近級差セ制冷 105 PV 0.1 × 0.1 ~ 4 × 4				i .	•								差电制度 25	
HgCdTe PC 0.25 × 0.25 180 3 ~ 5 4.6 > 10 ⁵ (0.5 ~ 5) × 10 ³ 1000 240 300 ~ 800 ~ 1 二级福差电洞冷 180 PV 0.1 × 0.1 ~ 4 × 4 5.3 1 × 10 ⁴ D ₂ = 1.2 × 10 ¹ 800 77 5 × 10 ² × 10 ³ < 1	HgCdTe	PC	0.25 × 0.25	180	3~5	4.2	>3 × 10 ¹	>5×10°	1000	300	100~300	~1		26
in5b PV 00.5~06.2 60 3~5.5 5.0 (1~5)A / W (0.5~3)×10 ¹⁰ 680 77 10~10 ⁶ <1 疾病柱瓦瓶帯金属所 105b PV 00.5~06 60 1~5.5 5.0 10 ¹ ~10 ⁴ (0.5~3)×10 ¹⁰ 1000 77 10 ¹ ~10 ⁶ <1 疾病柱瓦瓶帯金属所 105b PV 01~04.5 3~5 5±0.1 10 ³ ~10 ⁴ D ₁ = (0.5~1)×10 ¹¹ 1250 77 (1~30)K <1 20 20 20 20 20 20 20 2	HgCdTe	PC	0.25 × 0.25	180	3~5	4.6	>10,	$(0.5 - 5) \times 10^{9}$	1000	240	300~800	~1		24
InSb	InSb	PV	0.1×0.1~4×4			5.3	1 × 10 ⁴	$D_{s} = 1.2 \times 10^{11}$	800	77	5 < 103~ 108	<1		
InSb				60	3~5.5	5.0	(1~5)A/W		680	77	103~106	<1	玻璃柱瓦瓶带金属	屏蔽売 つフ
RgCdTe PC 0.2 × 0.2			7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7	60	1~5.5	5.0	10'~10"	$(0.5 - 2) \times 10^{10}$	0001	77	101~106	<1		
HgCdTe PC 0.2×0.2 8~14 ~10 (0.5~1.5)×10 ¹⁶ 1000 77 30~300					3~5	5 ± 0.1	103~104	$D_1 = (0.5 - 1) \times 10^{11}$	1250	77	(1~30)K	<1 20		
HgCdTe PC 0.2×0.2~0.4×0.4 8~14 10.5~14 10 ⁵ ×10 ⁵ D _s = (0.6~1)×10 ¹⁰ 900 10.5 20~50 0.01~1 HgCdTe PV Ф0.1~Ф0.5 8~14 11 10 ³ ×10 ⁴ D _s = (3.4)×10 ¹⁰ 900 77 20~200 10 ⁻³ HgCdTe PV 0.25×0.25 8~14 9~11 10 ³ D _s = (2.7.3)×10 ¹⁰ 1250 77 30~200 <0.1 接換柱形成 28 HgCdTe PV 0.5×0.5 180 8~14 ~10~12 (1~6)AW ⁻¹ D _s = (1.7.3)×10 ¹⁰ 630 77 50~1000 <0.02 20.02					8~14	~10			1000	77	30~300			
HgCdTe PV Φ0.1~Φ0.5 ** 8~14 11 10°×10° Dj = (3~4)×10° 900 77 20~200 10°3 HgCdTe PV 0.25×0.25 8~14 9~11 10° Dj = (3~4)×10° 1250 77 30~200 <0.1 玻璃柱瓦瓶 28 HgCdTe PV 0.5×0.5 180 8~14 ~10 12 (1~6)AW*1 Dj = (1~4)×10° 630 77 50~1000 0.2~0.02 玻璃柱瓦瓶 28 HgCdTe PV 0.5×0.5 180 8~14 ~10 Dj = (3~4)×10° 1000 77 50~1000 <0.01 金属柱瓦瓶 28 PbSnTe PV Φ0.1~Φ0.5 60 8~14 11.7 Dj = (1~3)×10° 1000 77 50~1000 <0.01 金属柱瓦瓶 28 PbSnTe PV Φ0.1~Φ0.5 60 8~14 11.7 Dj = (1~3)×10° 1000 77 50~1000 0.1					8~14	10.5~14	$(0.5-5)\times10^{\circ}$	$D_{r} = (3 - 4) \times 10^{10}$	2500	77	20~100			
HgCdTe PV 0.25 × 0.25 8 ~ 14 9 ~ 11 10¹ D∫ = (2 ~ 3) × 10¹¹¹ 1250 77 30 ~ 200 < 0.1 映解柱瓦版 28 HgCdTe PV (5 ~ 20) × 10⁻² 60 8 ~ 14 10 ~ 12 (1 ~ 6)AW⁻¹ D∫ = (1 ~ 4) × 10⁻¹ 630 77 50 ~ 1000 0.2 ~ 0.02 破離柱瓦版 28 HgCdTe PV 0.5 × 0.5 180 8 ~ 14 ~ 10 D∫ = (1.8 ~ 2.4) × 10¹¹¹ 1000 77 50 ~ 1000 < 0.01 金属柱瓦版 28 PbSnTe PV 0.1 ~ 00.5 60 8 ~ 14 11.7 D∫ = (1 ~ 3) × 10¹¹¹ 1000 77 > 30 0.2 PbSnTe PV 0.3 × 0.3 ~ 0.6 × 0.6 8 ~ 14 11.7 D∫ = (1 ~ 3) × 10¹¹¹ 1000 77 > 500 ~ 1000 0.1					8~14	10.5~14	$10^2 \times 10^2$	$D_{\star} = (0.6 - 1) \times 10^{10}$	900	105	20~50	0.01~1		
HgCdTe PV (5~20)×10 ⁻² 60 8~14 10~12 (1~6)AW ⁻¹ D; -(1~4)×10 ⁻¹⁰ 630 77 50~1000 0.2~0.02 破球柱形態 28 HgCdTe PV 0.5×0.5 180 8~14 ~10 D; -(0.8~2.4)×10 ⁻¹⁰ 1000 77 50~1000 <0.01 金属柱形態 28 PbSnTe PV 0.1~00.5 60 8~14 > 10 ² D; -(1~3)×10 ⁻¹⁰ 1000 77 >30 0.2 PbSnTe PV 0.3×0.3~0.6×0.6 8~14 11.7 D; -(1~3)×10 ⁻¹⁰ 900 77 500~1000 0.1			Ф0.1~Ф0.5	+	8~14	11	101×104	$D_j^* = (3 \sim 4) \times 10^{10}$	900	77	20~200	10-3		
HgCdTe PV (5-20)×10 ⁻² 60 8~14 10~12 (1~6)AW ⁻¹ D _s ² = (1~4)×10 ¹⁰ 630 77 50~1800 0.2~0.02 政務柱形版 28 HgCdTe PV 0.5×0.5 180 8~14 ~10 D _s ² = (0.8~2.4)×10 ¹⁰ 1000 77 50~1800 <0.01 金属柱形版 28 28 28 28 28 28 28 2					8~14	9~11	10,	$D_{s}^{*} = (2 \sim 3) \times 10^{10}$	1250	77	30~200	< 0.1	玻璃杜瓦瓶	28
HgCdTe PV 0.5×0.5 180 8~14 ~10 D; = (0.8~2.4)×10 ¹⁰ 1000 77 50-1000 <0.01 金属柱に瓶 28 PbSnTe PV 0.1~00.5 60 8~14					8~14	10~12	(1~6)AW-1	$D_{i}^{*} = (1 \sim 4) \times 10^{10}$	630	77	50~1000			
PbSnTe PV $0.1 - 90.5$ 60 $8 - 14$ $> 10^3$ $D_s^2 = (1 - 3) \times 10^{10}$ 1000 77 > 30 0.2 PbSnTe PV $0.3 \times 0.3 - 0.6 \times 0.6$ $8 - 14$ 11.7 $D_s^2 = 2.3 \times 10^{10}$ 900 77 $500 - 1000$ 0.1		•		180 -	8~14	~10		$D_{\star} = (0.8 \sim 2.4) \times 10^{10}$	1000	77	50~1000			
0.11 00 00 00 00 00 00 00				60	8~14				1000	77	> 30			<u> </u>
0 11 00 0 0 0 0 0 0 0 0 0 0 0 0 0 0					8~14	11.7		$D_{i} = 2.3 \times 10^{10}$	900	77	500~1000	0.1		
Ge ₁ Hg PC 0.5×0.5 $8 \sim 14$ 10.6 5×10^4 $\sim 10^{10}$ 800 30 $(50 \sim 100)$ k < 0.01	Ge₁Hg	PC	0.5 × 0.5		8~14	10.6	5 × 104	~ 1010	800	30	(50~100)k			

[KEY on next page]

KEY: 1 - name of detector 2 - operating mode 3 - area receiving light 4 - angle of field of view spectral interval 6 - peak value wavelength 5 - optical 6 - peak value wavelength 7 - response rate 9 - modulation frequency 8 - detection rate 10 - working 11 - impedance 12 - response time 13 - structural features 14 - quartz cover plate 15 - germanium flat window 16 - transistor 17 - fitted with Dewar glass flask 18 - immersion lens 19 - OD6 silicon dualsurface spherical lens, two levels 20 - thermocoupling electric cooling 21 - germanium immersion lens 22 - band immersion 23 - silicon dual-surface spherical lens 24 - two levels of thermocoupling electric cooling 25 - four-level thermocoupling electric cooling 26 - conditions of germanium immersion 27 - Dewar glass flask with metal shield in shell 28 - Dewar glass flask

(CCD) opened a route for the above-mentioned requirements, by installing the CCD onto an integrated module of the detector array. The CCD stores and processes the signal detected by the sensor. Without scanning, on seeing an object through a reflecting mirror the entire field scene can be observed in the same time period. The higher the number of detection elements, the higher is the sensitivity of the system, and the higher is the resolving power of the possible image output [8].

With respect to infrared imaging guidance of a focal-plane array, it improves the thermal sensitivity so that the detecting distance is extended and the discrimination rate is enhanced. With numerical processing by using an appropriate algorithm with thermal data from the array, the target signal can be provided and the target with the highest threat level can be determined. By designing a more effective optical system and adopting the advanced circuitry packaging, the size of the device can be reduced in order to reduce power consumption.

At present, the focal-plane array structure under development is divided into single-chip type or hybrid type. In the single-chip type, the detector and the CCD are fabricated on the same semiconductor material. They are unable to reach optimization together; frequently, optimization at one layer

TABLE 3. Performance Indicators of Typical Multielement Infrared Detectors

1 探测器名称	波长范围 (μm)	工作 方式	4 探測率 D _s (cm · Hz ^{1/2} · W ⁻¹) 19	上升时间 5 (μS)	近件尺寸 6 (μm²)	元件间隔 7 (μm)	8 _{元件数目} -13	9 制备方法	工作 温度 10k)	备注 11
PbS	1~3	PC	平均 2×10 ¹⁰		25 × 25	15	1024(多达 2000)	化学淀积 12	171	
PbSe	3~5	PC	(2~4)×10 ¹⁰		25 × 500	50	64(多达 140) 13	化学淀积 12	193	
InSb (平面镶嵌) 18	3~5	PV	$D_{SOUL}^{*} > 2 \times 10$		250 × 250	50	34(多达 556)13	离子注人 Zn 14	77	
InSb	3~5	PV	19	16	40 × 40	20	140		77	28
HgCdTe	3~5	PC	平均 D*(500k,20kHz,1)0.9×1010	最大 0.4	50 × 50	12.5	10~64		195	四级温差电制冷
HgCdTe	8~14	PC 1		最大3 16	50 × 50	12.5	48		77	
HgCdTe	8~14			缺大 0.4	50 × 50	12.5 或 18 15	50		> 65	36°视场 29
HgCdTe	8~14	PC	D*(500k,20kHz,1)0.9×1010]6	0.3	50 × 50	12.5	195(100 ~ 200)		77	60°视场 29
HgCdTe	2~12.5	PC	4×10 ¹⁰	0.4			20		77	$\lambda_{\rm P} = 11 \mu{\rm m}, {\rm Rv} = 1.5$ × $10^5 {\rm V} \cdot {\rm W}^{-1}$
HgCdTe	8~14	PC			40 × 40	8	180		77	
HgCdTe	8~14	PC	D*(10.5µm,5kHz,1)2×1010		50 × 50	25	10	扩散 21	77	180°视场 29
HgCdT¢	8~14	PV	$D^*(10\mu m, 1kHz, 1)6 \times 10^{10}$		100 × 100	25	80		77	30 ° 视场 29
HgCdTe	8~14	PV	NEP=6×10 ⁻²⁰ W·Hz ⁻¹	频啊 > 1.2Gc			4		77	4MHz 四象限外差, λ _p =1.06μm, 60°30 视场,战小 NEP<1. 10 ¹⁹ W·Hz ⁻¹ 3]
HgCdTe	8-14	PV	$D^*(10.5\mu m)1 \times 10^{11}$				20~200	外延 22 离子注人	77	战北/> 1000MHz 31
PbSnTe	8~14	PV	2 × 10,10		150 × 250	250	43 23	异质液相外延	77	RoAd2Ω · cm ⁻²
	8~14	PV	2.3 × 10 ¹⁰		6 × 10 ⁻³ mm ²		18	24 液相外延	80	29 2π 视场, 300K 背景 32
in-PbSnTc 17	8~14	PV	3.5 × 10 ¹⁰		7.33 × 10 1mm ²		25 25	分子束外延	77	λ _p =10μm, 180°视场
肖特基势垒			2.2×10 ¹⁰ (9.8µm)							
P-PbSnTe/	8~14	PV	1 × 1011(4.6µm)				12(各.6) 26		80	
P-PbTe	3~5									
Ge:Hg	8~14	PC			100 × 100	50	60(多达 180) 27		35	

[KEY on next page]

KEY: 1 - name of detector 2 - wavelength interval 3 - operating type 4 - detection rate 5 - rise time 6 - dimensions of element 7 - spacing of elements 8 - number of elements 9 - preparatory method 10 - working temperature 11 - remarks 12 - chemical deposition 13 - as many as 14 - ion implantation 15 - or 16 - maximum 17 - Schottky potential barrier 18 - flat fitted 19 - average 20 - frequency response 21 - diffusion 22 - epitaxial 23 - heterogeneous liquidphase epitaxy 24 - liquid-phase epitaxy 25 - molecular-26 - (six each) 27 - as many as beam epitaxy 28 - four levels of thermocoupling electric cooling 29 - field of 30 - four-quadrant epitaxy 31 - field of view, minimum 32 - background

leads to reduced performance at another layer. In the hybrid type, the detector array and the CCD layer can reach optimization separately. Then interconnections are made. In the present status, the hybrid structure is more emphasized than the single-chip structure with faster development. Generally, hybrid integration is made with photovoltaic detectors and Si-CCD. Fig. 10 [9] shows a typical structure of the hybrid type infrared CCD with a combination of photovoltaic HgCdTe detector and a Si-CCD.

Under development is another method of realizing the focalplane array; this is the Z-technique [10, 11]. Several ceramic chips with lead wire patterns overlapping each other involve internal circuits in the ceramic chip. The input signal of the sensor is sent by a lead wire according to the specified routing to the analog signal processor, and multichannel transmitter integrated circuit, installed on the ceramic chip. sensor is installed on the edge of the ceramic chip layers. This structure provides wide-open prospects for advanced signal processing. If the silicon integrated circuit replaces the ceramic chip, the design can be more compact; the output of each row of detection elements is directly coupled onto the input terminal of the silicon chip. The position of the silicon chip is vertical to the detection element chip (in the Z-plane).

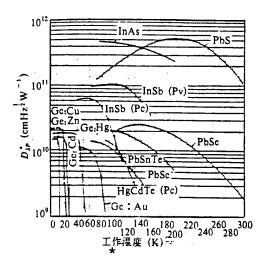


Fig. 7. Detection-temperature property of
typical infrared detectors
KEY: * - working temperature

chip can process a signal, and also cancel out jamming. The output of the processor chip is fed to the second level processor, whose position is face to face and parallel with the array chip (on the XY-plane). Thus, very high densities can be realized by keeping the processor physically closer to the detection element. However, costs under this method are very high.

To enable a focal-plane array to be applied to missiles and other systems in large lots, cutting production costs is very important. Increasing the end product yield and reducing the substrate dimensions are the key to lower costs. In the Pave Pace project in the United States, the main target is to reduce the production costs of the focal plane array from 60,000U.S. dollars for each focal-plane array at present, to less than 1,000U.S. dollars.

The Texas Instruments Corporation (TI) in the United States

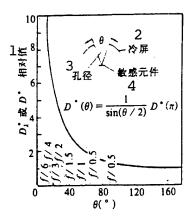


Fig. 8. Relation between detection rate and field-of-view angle of infrared detectors KEY: 1 - relative value 2 - cooled shield 3 - aperture 4 - sensitive element

has attempted to comprehensively apply the silicon substrate manufacturing technique, as well as the experience and production equipment in order to mass produce low-cost focal-plane arrays. Fig. 11 shows the production process. By adopting the interconnection method as shown in Fig. 12, of the vertical integrated metal insulated semiconductor (VIMIS) to interconnect the infrared detector and the integrated circuit processor, by keeping good contacts.

In another production method, as adopted by the Rockwell Corporation in the upgrading project of Pace I, the focal-plane array production line with sapphire substrate. In the production process, the HgCdTe grown with extension at a low temperature adheres to the substrate so as to realize high uniformity and stable sensitivity. In addition, through indium contact pads, each detection element is directly connected with the silicon integrated chip, as shown in Fig. 12. It was stated that the

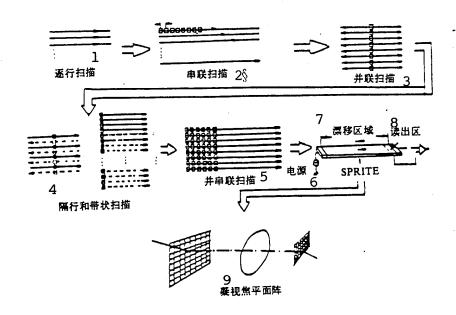


Fig. 9. Schematic diagram of single-element scanning of staring-type focal-plane array development KEY: 1 - line-by-line scanning 2 - serial scanning 3 - parallel scanning 4 - scanning at alternate lines and zone scanning 5 - parallel and serial scanning 6 - power supply 7 -drift region 8 - readout zone 9 - staring-type focal-plane array

density is the highest with this packaging, along with good interconnections and without shielding against incident thermal radiation.

TABLE 4. Main Properties of SPRITE Detector

材料 17	n-HgCdTe	n-HgCdTe
探测元数量 1	8	. 8
条形片子长度(μm) 2	700	700
光敏面(瞬时)(μm²) 3	62.5 × 62.5	62.5 × 62.5
工作波长(μm) 4	8~14	3~5
工作温度(K) 5	77	190
致冷方式 6	节流或热机 15	温差电 16
偏置场强(V・cm ⁻¹) 7	30	30
迁移率(cm²·V⁻¹S⁻¹) 8	390	140
像元速率(像元・S ⁻¹) 9	1.8 × 10 ⁶	7×10^5
探测元典型阻值(Ω) 10	500	4.5×10^3
每探测元功耗(mW) 11	9	1
探測器总功耗(mW) <u>12</u>	< 80	< 10
平均 D*(500K,20kHz 1, 62.5× 13	>11×10 ¹⁰	$(4 \sim 7) \times 10^{10}$
62.5 μ m ²)(cm • Hz ^{1/2} • W ⁻¹)		
响应率 R(500K, 62.5×62.5µm²)	6×10 ⁴	
$\times (\mathbf{V} \cdot \mathbf{W}^{-1})$		

KEY: 1 - number of detection elements 2 - length of strip-shaped foil 3 - photosensitive surface (instantaneous)
4 - working wavelength 5 - working temperature 6 - cooling method 7 - bias field intensity 8 - transfer rate 9 - speed of picture element (picture element S⁻¹) 10 - typical resistance of detection element 11 - power consumption for each detection element 12 - total power consumption of detector
13 - average 14 - response rate 15 - throttling or heat engine 16 - thermocoupling electricity

In the Pace I stage, up to now the largest hybrid focal-

plane array that can be produced is the 256×256 $40 \mu m$ -element detector. It is estimated that if a 5.08 cm substrate is used, calculated on the basis of a 0.5% yield rate at present, daily

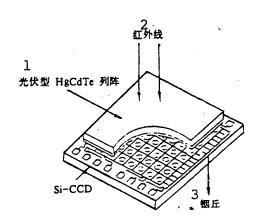


Fig. 10. Hybrid type infrared CCD structure KEY: 1 - photovoltaic type HgCdTe array 2 - infrared array 3 - indium pad

output should be 5000substrates in order to satisfy the annual demand of half a billion picture elements in the nineties, corresponding to 9U.S. dollars per picture element. The goal of the Defense Strategic and Tactical Reproduction (DSTAR) of the Defense Advanced Research Project Agency in the United States is to use 7.2cm substrates for a yield rate of up to 15%, and the cost of each picture element to be lowered to 5cents. Thus, the daily output of nearly 50substrates can satisfy the abovementioned goal. Generally, with respect to infrared focal-plane arrays, manufacturing costs and technology are always problems of concern to the parties involved.

2. Development of infrared signal processing [12, 13]

In the nineties, with development of tracking reconnaissance toward real-time capability, infrared sensors used for detection and tracking will have very high data rates. Therefore, this requires adopting high-speed hardware with parallel processing, with adoption of computer graphic signal processing, including preprocessing (graphics enhancement and division) video signal

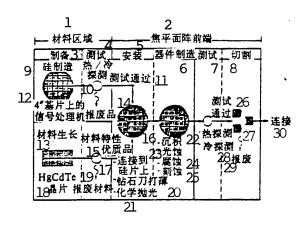


Fig. 11. Focal-plane array fabrication flow chart at TI Corporation KEY: 1 - region of material 2 - front end of focal-plane array 3 - preparation 4 - measurement and testing 5 - mounting 6 - fabrication of device 7 - measurement and 8 - cutting 9 - silicon fabrication testing 10 - hot/cold detection 11 - passed measurement 12 - signal processor on 4-inch substrate 14 - reject product 13 - growth of material 15 - material properties 16 - high-grade product 17 - interconnections to silicon chip 18 - chip 19 - reject material 20 - thinning with diamond 21 - chemical polishing cutter 23 - optical etching 22 - sedimentation 24 - corrosion etching 25 - carved etching 26 - passed measurement and testing 27 - hot detection 28 - cold detection 29 - reject 30 - interconnections

frequency processing (target feature extraction and classification of decision-making), and target position data processing (to have control command). As required by preprocessing, the computational rate is 5×10^6 operations per second. Development of very-high-speed integrated circuits

(VHSIC) provides the possibility of computations at high density and superhigh speeds. It is estimated that by the end of the century, the silicon integrated circuit will have matured in

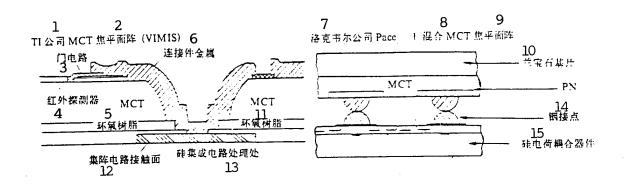


Fig. 12. Schematic diagram of focal-plane array interconnections KEY: 1 - TI Corporation 2 - focal-plane array 3 - gate circuit 4 - infrared detector 5 - epoxy resin 6 - connector metal 7 - Rockwell Corporation 8 - mixed 9 - focal-plane array 10 - sapphire substrate 11 - epoxy resin 12 - contact surface of integrated-array circuit 13 - processing site of silicon integrated circuit 14 - indium contact pad 15 - silicon charge-coupled device

technology; the number of transistors contained on each piece of integrated circuitry will be tens of millions, even up to 100million, 100times the present number of transistors.

With respect to the infrared focal-plane array sensors, adoption of advanced graphic signal processing technique is the key to achieving intelligent sensors [10]. Further development of the signal processing technique, including the matching of each detection element on the focal-plane array, uniform response on about 1% of the terminal wavelengths, with compensation attained from optical filtering and signal processing.

Development of VHSIC will further promote upgrading of the information processing technique. With the development of the advanced infrared focal-plane array of very high density and very

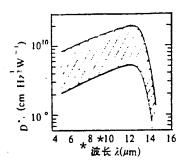


Fig. 13. Optical spectral detection rate of HgCdTe detector (working temperature: 77K, 10° to 130°, field of view, 300K background, photosensitive area 4x10⁻⁵ to 4x10⁻²cm²)
KEY: * - wavelength

high picture element number, one can determine the upgrading of the parallel techniques. For example, the SIMD (single-instruction/multiple data) configuration already used in focal-plane arrays, this parallel technique will move toward superhigh-performance and highly compilation simplification. In addition, with the development toward parallel-coupling with the local number of processing elements and a large number of high-speed data memories, this will provide more conventional computational capability for real-time processors of focal-plane graphics.

3. Superconducting infrared focal-plane array of the future [14]

The development of superconducting technology makes the application possible on space-based brand-new sensor technology

based on superconductivity. The advantages of using the superconducting circuits of low-heat dissipation used in focal-plane arrays are as follows: heating is no more the principal restraining factor in chip-making of high processing capability.

TABLE 5. Major Performance Indicators of HgCdTe Detectors

参数名称]	光电导型 2	光伏型 3
工作温度(K)	77	77
时间常数(μs) 5	~1	< 0.1
光谱范围(μm)	8~14	8~14
/ 探測率 D ₁ (cm・Hz ^{1/2} ・W ⁻¹)	> 1 × 10 ¹⁰	$> 1 \times 10^{10}$

KEY: 1 - name of parameter 2 - photoconductive type

3 - photovoltaic type 4 - working temperature

5 - time constant 6 - optical spectral interval

7 - detection rate

It is very easy to carry out analog-to-digital conversion by using superconducting circuitry. Digital processing related to target motion can be handled with chips. All such digital output can be mixed into a single electro-optical modulator, to modulate the fiber optic light beam and to communicate with the outside world. It is estimated that the power consumption in accomplishing the above-mentioned functions is only a fraction of the semiconductor focal-plane array at present. Moreover, it is possible to utilize the available superconducting manufacturing technique to make large focal-plane arrays of several square inches. This will be revolutionary progress with respect to the technology of infrared focal-plane arrays.

IV. Several Types of Infrared Detectors with Development Prospects [15-20]

To explain the development of infrared detection and tracking technology to the next step, below we present HgCdTe, platinum silicide and bichromatic detectors with

TABLE 6. Performance Indicators of Bichromatic Detectors

测试条件 1		
黑体温度 2	500K	
背景温度 3	300K	
元件温度 4	77K	
辐射通量密度 5	2μW / cm ²	•
带 宽 6	6Hz	
窗口材料 7	IRTRAN-2	
探測器间隔 8	0.005"	
结果 9	PbSnTe	InSb
探测器面积(cm²) 10	0.02	0.045
开路电压(mV) 11	11	98
短路电流(μA) <u>12</u>	6	2.2
零偏压阻抗(kΩ) 13	3.2	530
零偏压电容(pF) 14	940	32
峰值响应率(×10 ⁵ V/W)	0.046	6.25
$D^{\bullet}(500,900,1)(\times 10^{10} \text{cm} \cdot \text{Hz}^{1/2}/\text{W})$	0.41	1.7
$D^*(\lambda_p, 900, 1)(1 \times 10^{10} \text{cm} \cdot \text{Hz}^{1/2} / \text{W})$	1.45	8.5

KEY: 1 - measurement and test conditions 2 - blackbody temperature 3 - background temperature 4 - element temperature 5 - density of radiation flux 6 - bandwidth 7 - window material 8 - detector spacing 9 - results 10 - detector area 11 - open-circuit voltage 12 - short-circuit current 13 - impedance at zero bias 14 - capacitance at zero bias 15 - peak value response rate

development prospects in miliary applications.

1. HgCdTe infrared detectors

Because of the physical properties of HgCdTe material, this

is the best material from which to make infrared detectors, with the most applications in the 8 to $12\mu m$ waveband. Table 5 and Fig. 13 show its principal performance indicators and parameters.

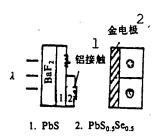


Fig. 14. Structure of bichromatic detector KEY: 1 - aluminum contact pad 2 - gold electrode

The advantages of HgCdTe are as follows [15]: (1) the material can be used for infrared detection in various response wavebands; (2) high response rate; (3) large photoelectric gain factor; (4) high quantum efficiency; and (5) high sensitivity. However, there are disadvantages to HgCdTe: cooling is necessary; homogeneity of picture elements is not sufficiently ideal; and higher production costs. After many years' research, the HgCdTe material fabrication technique has grown in maturity, with the properties of crystal, structure, and physical performance being fairly well understood. Not only have good-performance singleelement and line-array detectors been fabricated, but HgCdTe is also the most important material in developing focal-plane The most 128x128intermediate wave HgCdTe array made by Rockwell Corporation has a quantum efficiency of 64%; by using $60\mu m$ -diameter picture elements, helicopters as far away as 8kmhave been detected. At night time, nonoperating armored vehicles 4.5km away can be detected.

2. Platinum silicide infrared detectors [16]

Among the more materials that are more sensitive to infrared radiation, besides HgCdTe with the most applications in the focal-plane array, another very attractive material for stare-

TABLE 7. Performance Indicators, Application Status and Key Techniques of Some Foreign Infrared Imaging Guided Missiles

海别1	型 号 2	性能指标 3	应用现状 4	关键技术 5
	-幼育* <u>10</u> AGM-65D	20 电视、红外成像、激光三种 导引头、对付地面各种目标	32 1974 年研制、1983 年具有 作战能力、1993 年大批装备	理,小型光机扫描系统 43
美	"拉姆" RAM 11	雷达与红外双模被动导引头, 对付反舰导弹 21		被动雷达导引头与被动红 外天线通用复合 44
6	12 "海尔法" AGM-114A	激光半注动、红外成像和射 频一红外三种导引头对付装 甲目标 22	1972 年研制,1982 年投产, 1984 年大批装备 34	激光指示和通用导弹控制 系统 45
国	"黄蜂" Wasp 13	毫米波雷达与红外成像制导,攻击机动装甲 23	1982 年 研 制 , 1986 年 生 产 , 1988 年 装备 35	识别技术,数据处理,天 线 <u>罩</u> 46
	-战斧" 14 BGM-109A	红外成像制导,巡航高度 1.2km 24	1972 年研制、1982 年生产、 1988 年装备 36	图像处理,大规模集成计 算机 47
	"坦克破坏者" 15	凝视焦平面成像,攻击坦克、 直升机 25	1981 年研制, 1982 年试验, 尚未装备 37	凝视焦平面成像技术
	"先进巡航导弹"16	红外、毫米波制导26		
跨国 7	"ADATS"防 空、"反坦克导弹" 17	雷达、电视、光学、红外成像及激光制导对付飞机、装甲、坦克 27	1979 年研制, 1981 年试验, 1984年后装备 38	两用精度协调,两用战斗 部协调 49
8苏联	SA-9	被动红外成像制导 28	已装备 39	
美国 6	"中程空空导弹" (AMRAAM) 18	凝视焦平面 4096 元和 1024 × 1024 元制导 29	40 1983 年论证,90 年代装备	多元探测,多路信息传输、处理和计算机
英国 9	*近程空空导弹* (ASRAAM)]9	凝视焦平面 1024×1024 元 制导 30	40 4 <u>1</u> 1983 年论证,90 年代装备	多元探测、多路信息处理 和计算机 51
8苏联	AS-13	被动红外成像制导 31	在研制中 42	

[KEY on next page

2 - model number 3 - performance indicator KEY: 1 - nation 4 - application status 5 - techniques 6 - United States 7 - multinational category 8 - Soviet Union 9 - United 10 - Bullpup 11 - RAM 12 - Hellfire 13 - Wasp 14 - Tomahawk 15 - Tank destroyer 16 - Advanced cruise missiles 17 - air defense and antitank missiles 18 - intermediate air-to-air missiles 19 - short-range air-toair missiles 20 - three guidance heads of television, infrared imaging and laser to deal with various ground targets 21 - radar and infrared dual mode passive guidance heads to deal with antiship missiles 22 - three kinds of guidance heads: laser semiactive, infrared imaging, and radio-frequency-infrared, to deal with armored targets 23 - millimeter radar and infrared imaging guidance to attack motorized armor 24 - infrared guidance, cruise height at 1.2km 25 - staringtype focal-plane imaging to attack tanks and helicopters 26 - infrared and millimeter wave guidance 27 - radar, television, optics, infrared imaging, and laser guidance, to deal with aircraft, armor, and tanks 28 - passive infrared imaging 29 - staring-type focal-plane 4096-element, and quidance 1024x1024-element guidance 30 - staring-type focal plane 1024x1024 element guidance 31 - passive infrared imaging 32 - development in 1974, combat capability in 1983, quidance and deployment in large numbers in 1993 33 - development in 1979, test flight in 1980, scheduled to deploy after 1986 34 - development in 1972, production in 1982, and deployment in large numbers in 1984 35 - development in 1982, production in 1986, and deployment in 1988 36 - development in 1972, production in 1982, and deployment in 1988 37 - development in 1982, experimentation in 1982, not yet deployed 38 - development in 1979, experiments in 1981, and deployment after 1984 39 - already deployed 40 - verification in 1983, deployed in the nineties 41 - verification in 1983, deployed in 42 - in development 43 - multielement infrared the nineties imaging technique, graphics processing, small optical device scanning system 44 - passive radar guidance head, and passive infrared antenna applications 45 - laser indication and general missile control system 46 - discrimination technique, data processing, and antenna dome 47 - graphics processing with large-scale integrated computer 48 - staring-type focal-plane imaging technique 49 - dual-usage precision coordination and dual-usage warhead coordination 50 - multielement detection, multichannel information transmission, processing and computer 51 - multielement detection, multichannel data processing and computer

type material is platinum silicide.

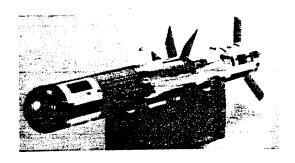


Fig. 15. AAWS-M guided missile

Platinum silicide detectors rely on photoradiation technology to detect infrared photons, which collide on the platinum electrode of the detector, stimulating electrons entering the silicon chips. Beneath the platinum silicide thin layer, the silicon integrated circuit is a detector array with multiple converters that collect signal charges from each detection element, to convert the charges into a video-frequency readout signal. This signal should be amplified with synchronizing pulses to be able to display the infrared pictures on a monitor.

In fabricating the platinum silicide detector, first the platinum is vaporized at 2000C for deposition on a silicon integrated circuit several feet away. When platinum arrives at the silicon chip, the metal vapor temperature has been reduced to hundreds of degrees centigrade. However, as mentioned above, the molten platinum silicide does not cover the entire detector surface, because there are silicon processing circuits around the rectangular platinum silicide.

Production costs are the key factor determining the success of new detector models. One of the principal reasons for

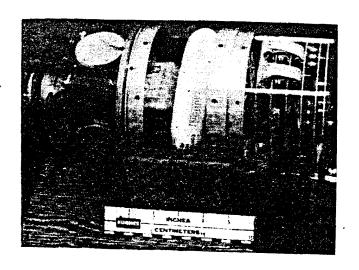


Fig. 16. FOG-M infrared guidance head

platinum silicide being highly regarded is its low costs, although its sensitivity is not as good as that of HgCdTe. By coating 3 to 5µm in the intermediate infrared window, the cost is considerably lower than that of HgCdTe. Every picture element with HgCdTe detection costs, in terms of dollars, while a picture element of a platinum silicide detector costs only cents. At present, some companies and research centers are developing a detector array chip containing 250,000 to 300,000picture elements, in two specifications of 512x512, and 486x640picture elements. The 1024x1024 elements, as an array including more than 1million picture elements, is also expected to become available several years in the future.

The quantum efficiency of platinum silicide is not as good as HgCdTe, but its homogeneity is better. Moreover, only 16% of the picture elements covered by a platinum silicide detector should be improved; the other portions are covered by the electronic readout data circuits. Thus, the problem can be

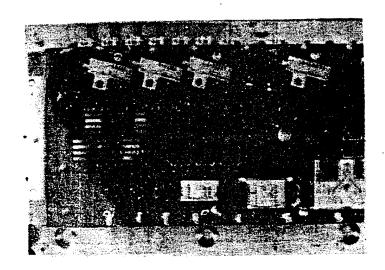


Fig. 17. There are 15modules in focal-plane array of AOA sensor; each module has four sensor chip arrays, including 640detection elements. One module in the picture cannot be seen

solved by fabricating a hybrid device. In other words, the detector and readout equipment are separated by two silicon chips, which are interconnected with indium contact pads with ceramic chip packaging. In this way, the interstitial coefficient at the top of the silicon chip can be as high as 95%, so that almost the entire surface of the picture elements have a detector function. The multiple conversion circuit is placed on the second block*, thus becoming more complicated in order to process signals with very wide dynamic variation range. With shrinkage of the silicon integrated circuit and improvements in processing, at present it is possible to put the multiple converter and detector on the same piece of material, capable of simultaneously attaining high interstitial coefficient.

One property of the platinum silicide detector that is noteworthy is to separate the backgrounds between ${\rm CO}_2$ and the

infrared energy that can be radiated. This has important military value, because ${\rm CO}_2$ gas is included in the exhaust of aircraft and missile engines.

3. Gallium arsenide detectors [17, 18]

Gallium arsenide infrared detectors are noteworthy. Together with HgCdTe, gallium arsenide has the same property of detecting long-wave infrared.

At Bell Laboratories in the United States, an entirely-new method was used in fabricating gallium arsenide infrared detectors. The entire physical detection process is also different. Gallium arsenide is used to replace HgCdTe, and a sandwich structure, which the HgCdTe detector lacks, is applied. By using silicon doping, 40Angstrom thick cavities are formed to provide a mechanism for detecting photons. Cavities are separated by the gallium aluminum arsenide potential barrier; the substrate is a piece of gallium arsenide. This new infrared detector is modulated by changing the cavity depth and constituents of the potential barrier, capable of detecting wavelengths between 6 and $10\mu\text{m}$. Up to the present time, the most effective detectors that have been fabricated are used in detecting $8-\mu m$ wavelengths. The advantage of this approach is that the high speed gallium arsenide field-effect transistor (FET) can be integrated with the infrared detector onto a single piece of material. However, the HgCdTe detector requires the use of a silicon semiconductor exclusively to fabricate circuitry. Since a single piece of gallium arsenide material can be used to fabricate electronic circuitry and detection devices, this is very useful in fabricating focal-plane array detectors. addition, the gallium arsenide focal-plane array is possibly ten times cheaper than the HgCdTe.

Besides, integrated circuitry using gallium arsenide is five

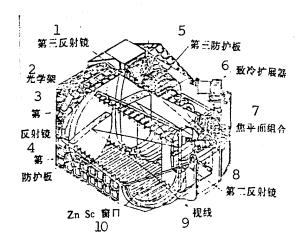


Fig. 18. Three-surface molten quartz reflecting mirror to reflect photons onto a focal-plane complex, 15inches long, in AOA sensor. The complex has 38,400detection elements, with low-temperature cooling.

KEY: 1 - third reflecting mirror 2 - optical frame 3 - first reflecting mirror 4 - first protective plate 5 - third protective plate 6 - cooling expander 7 - focal-plane complex 8 - second reflecting mirror 9 - line of sight 10 - ZnSc window

times faster in operating than when using silicon, with a wide adaptation range and low power consumption, so that gallium arsenide is an indispensable material for supercomputers and optical communications. Moreover, the properties of gallium arsenide are higher with respect to cosmic rays and nuclear radiation; this is an ideal material for space military applications. Infrared sensors used in ballistic missile defense (BMD) systems should have the capability of withstanding intense radiation from nuclear blasts far away. The present infrared detectors are fabricated from thicker silicon and HgCdTe

materials, since these materials are easily damaged by nuclear blasts. Gallium arsenide detectors can meet the requirements of withstanding radiation in the BMD system.

However, like HgCdTe, this new type of detector can operate only after being cooled down to 77K. Liquid nitrogen is required for cooling. There are some differences from monolithic silicon elements, in that defects easily occur internally during crystallization, thus bringing difficulties to the high integration of integrated circuitry. Further research is required to overcome this difficulty.

4. Bichromatic detectors [19]

With advances in precision guidance, it appears very important to measure and compare two or more wavebands in the infrared spectrum. The so-called bichromatic or multichromatic detector involves using two or more responses in different wavebands to make a sandwich structure type detector, capable of simultaneously detecting radiation from two or more wavebands. After the seventies, bichromatic detectors, such as InSb/HgCdTe (1 to 5 μ m)/5 to 15 μ m), and the InSb/PbSbTe (1 to 5 μ m/5 to 12 μ m), were developed in the United States with mature preparatory techniques. Table 6 shows the principal performance indicators.

Fig. 14 shows a new bichromatic detector structure developed for the U.S. Navy. The unique features are intense display signals and good discrimination. The key technical process is to use vacuum vaporization to grow a 5 μ m-thick epitaxial p-type PbS layer 5 μ m on a transparent BaF₂ substrate, then a stainless steel foil cover plate is deposited over half the Pb layer with a 5 μ m-thick epitaxial layer of PbS_{0.5}Se_{0.5}P. On two epitaxial layers line-array a circular-shaped Pb electrode is vaporized, thus fabricating a bichromatic photovoltaic detector. The goal in

this research is to obtain, finally, a multichromatic focal-plane detector.

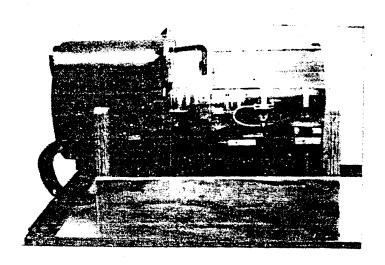


Fig. 19. Built by Martin-Marrieta Corporation, guidance head used for SBI, including high-speed imaging processor and focal-plane array bichromatic sensor

V. Applications of Infrared Detection and Tracking Technique [20-32]

1. Applications in tactical missiles

Infrared detectors are widely used in weapon guidance; up to now, infrared guidance has been used in 60 to 70missile models.

In combat, infrared guided missiles demonstrate precision In the past, for more than a decade more than 90% of known aircraft losses in wars were downed with infrared missiles In particular, infrared imaging guidance technology greatly upgrades weapon performance. Table 7 lists performance indicators of some infrared imaging guidance missiles made abroad, together with application status and key techniques [21]. The first finalized models of an infrared imaging missile, the Bullpup AGM-65D and AGM-65F in the United States, have been in lot production, deployed among troop units. However, missile improvements are still continuing. At the Hughes Corporation, development is underway on computer artificial intelligence and expert systems of graphic processing techniques by extracting information from graphics. In other words, a large-capacity memory is installed in a missile to store detailed information on the target selected, the scanning environment during missile flight, and an automatic search and retrieval database in order to upgrade missile performance.

Bichromatic guidance technique is one of the present important direction of development in precision guidance munitions. At present, more than ten type of bichromatic guidance cannon rounds and guided missiles have been developed, such as the SA-13 of the former Soviet Union, the improved Stinger version of the United States, and the TAN-SAM made in Bichromatic guidance utilizes the frequency spectrum difference between a target (such as an aircraft, missile, or tank) and the background (such as sky, sea surface, and ground objects), as well as various kinds of jamming. In the case of the U.S. Stinger missile, CdS/InSb (0.3 μ m/3 to 5 μ m) is employed, by using an $0.3\mu m$ (the peak scattering density value of sunlight) ultraviolet detector to distinguish between the target and the sky background, and a 3 to $5\mu m$ infrared detector to detect and track thermal energy radiated by aircraft (the peak value of radiation lies between 4 and $5\mu m$). Development is underway, in

the United States, on a bichromatic imaging system in the 3 to $5\mu m$ and the 8 to $13\mu m$ wavebands. The goal is to develop an imaging guidance head by using a 128x128bichromatic array.

An important technique of infrared sensors in order to achieve innovation in the future progress of infrared weapons is the infrared focal-plane array [10]. Beginning with recent development, the new-generation tactical missiles will apply the focal-plane array technique. For example, the portable advanced antitank weapon system--intermediate type (AAWS-M) of the U.S. Army, the fire-and-forget missile [15]) applies a 64x64 staring-type focal plane array. The thermal imaging instrument of the command launch unit (CLU) applies a 241x1scanning focal plane array for a wide field of view.

Besides the application of long-wave HgCdTe scanning and staring-type arrays by the AAWS-M, the advanced missile systemheavy version (AMS-H) and the over-the-horizon air defense/antitank missile (FOG-M, also called the fiber-optical guided missile), will also apply the focal-plane imaging guidance head. What has been adopted by the AMS-H is the long-wave HgCdTe 128x128 staring-type array. However, the FOG-M will plan to adopt the intermediate platinum silicide staring-type array. Previously, a platinum silicide staring-type focal-plane array was developed at the Hughes Corporation, for the IIR quidance head (Fig. 16) of the FOG-M, with the adoption of a 256x256structure, for a total of 56,536detection elements. Later, the company developed a larger 244x244staring-type array. Since the 114-pound FOG-M can contain more electronic equipment, the larger the array, the better the resolving power of video frequency that can be provided.

2. Applications in SDIO program

One of the 15 important experimental projects in the SDI program is research on long-wave infrared sensors [25].

Consideration was given to installing such sensors in the Space Surveillance and Tracking System (SSTS) satellite to detect reentry ICBM nose cones. Developed under the SDI program, the long-wave infrared sensors have successfully tracked the targets launched from the Vandenburg Air Force Base. The present longwave infrared detection technology has the capability of detecting, from 1000nautical miles away in cold outer space, a target with heat emitted by a human body. The long-wave infrared detection technique can be used not only to detect and track terminal reentry nose cones, but also can detect and track targets in the unpowered flight stage above the atmosphere, in order to gain precious early warning time for terminal defense. In the SDI program, nearly 1000early warning satellites primarily adopt the long-wave focal-plane array. In development at the present, the dynamic energy interceptor missile will also adopt the infrared detection and automatic tracking technique. be seen that the infrared focal-plane detection and tracking techniques have important standing in the SDI program [24].

(1) Satellite-borne infrared sensors [23]

Infrared sensors carried by infrared early warning satellites can detect missiles in the boost-phase. The maximum sensitivity is concentrated in the 0.3µm wave window. By using the 3 to 5µm intermediate wave window, although the energy liberated by the booster rocket is less than 1.5µm at 3.5µm, yet the absolute energy is large; therefore, within tens of seconds (50s, in the case of the United States) after launch of a strategic missile, the satellite-borne sensor can detect with 30min of early warning time for antimissile defense. Since 1971, satellite-borne infrared sensors of the United States have detected more than 1000launches of ballistic missiles by the Soviet Union, China, and France.

(2) Aircraft-borne infrared sensors

There is good detection capability of aircraft-borne infrared sensors used in detecting ballistic missiles. The U-2 high-altitude reconnaissance craft carried infrared heat sensors for the emergency early warning system. The infrared telescope has a field of view -2 to 20° , wavelength between 8 and $12\mu\text{m}$, aperture 20.3cm, and focal length 91.4cm, capable of seeing targets in the upper atmospheric layers. To enhance the aircraft-borne early warning capability, the Air Force made improvement in stages in order to have 125,000array elements, and an action distance as far as 4775nautical miles, capable of detecting various ICBM models in the boost phase.

Besides, the airborne optical apparatus (AOA) under development in the United States will also be used in BMD [25]. In this case, the infrared sensor is carried on aircraft to supplement ground radar in missile defense. The focal length set for the sensor is 15inches, including 60detector chips, and 640infrared detection elements for each array. With indium, the detector connects with a preamplifier having the same number of detection elements to amplify the signal slightly before entering the readout circuit. 38,400gallium-doped silicon detection elements are installed on 15modules, as shown in Figs. 17 and 18, in order to assess the performance in tracking and detecting missiles.

(3) Infrared guidance head of interceptor

As indicated by space-based interceptor (SBI) experiments in the SDI program [26], the SBI technique can be used to intercept ballistic missiles in the boost phase. Fig. 19 shows the guidance head of the staring-type focal-plane array of SBI, capable of discovering and locking on the target. As revealed in experiments, the SBI sensors, signal processors, and software can process simultaneously large amounts of signals in order to

discriminate the very bright exhaust plume from a solid-state rocket engine from the missile proper.

Under development at present is the light extra-atmospheric projectile (LEAP) project [27] with infrared homing. The LEAP project will be used to track and intercept targets in the boost phase, as well as in the late and intermediate phases. This project also includes the infrared imaging sensor electronic equipment, inertial sensor equipment, thrust-vectoring sensor, thermocoupling battery set, and the command data link, used in the kinetic kill system. The terminal homing sensor guidance system and the thrust-vectoring engine controlled by the onboard missile computer can ensure target interception. This year's experiment will verify the performance of 16,384elements of a 128x128 HgCdTe focal-plane array [28].

3. Applications in the Gulf War

In local wars in recent wars, the infrared early-warning satellites and early warning reconnaissance aircraft have been important means of detection and reconnaissance [29]. In this Gulf War, three early warning satellites were used. During limited visibility, or with camouflage colors painted on launchers leading to difficult-to-discriminate situations, the KH-12 Keyhole satellite 850km from the ground can utilize image forming and infrared scanning methods to detect heat energy from fired missiles in order to find their location. In addition, this satellite is responsible for photographing crisply clear pictures such as the serial number on the missile launcher.

Since the SCUD missile launchers of Iraq are mobile, sometimes reconnaissance satellites are not able to precisely discover their tracks. The GR1 Hurricane reconnaissance of the U.K. Royal Air Force exploited important functions [30]. At the forward undernose surface of the GR1, an infrared detection system is installed, capable of precisely detecting the tracks of

the SCUD, to inform friendly aircraft for attack. In the second round attack against Iraq, the GR1 successfully smashed six missile launchers.

TABLE 8. Optoelectronic Guided Weapons Emerging in the Gulf War

国别 1	名 称 4	5 类别	6 制导体制
多 闰 3 队	BGM-109C"战斧" 7	17巡航弹	## \- ## #\$ III # 1 : 60 /d = 2 //6 1 m 6 1
	"幼育"AGM-65D 8	18 空一地	红外成像制量 23
	· 轮剑 · 9	19防空弹	光学/电视或雷达跟踪+无线电指令
	"幼畜"AGM-65A / B	18空-地	电视制导 25 24
	"罗兰特 II" 10	20地一章	光学跟踪-无线电指令 26
	-尼刺POST 11	20地-空	41 01 mt 01 11 (2 1 40)
	-西北风 12	20地-空	11. 年 · 東 · 八 · 三 · 三 · 三 · 三 · 三 · 三 · 三 · 三 · 三
	"啊尾蛇"AIM-9系列 13	21 空-空	红外制导 29
	"海尔法"反坦克导弹 14	18空-地	激光半自动 30
	GBU-15. AGM-130	18空-地	中報制品
	宝石路、AS.30 15	18空-地	25 激光制导炸弹 31
伊 3	SA-9、SA-13 等 16	20地-空	红外制导 29
军	SA-14	20 地-空	激光驾束 32

[KEY on next page]

KEY: 1 - nation 2- multinational troops 3 - Iraqi troops 4 - name 5 - category 6 - guidance system 7 - Tomahawk 8 - Bullpup 9 - Lightsword 10 - Roland 11 - Stinger 12 - Northwesterly 13 - Rattler, AIM-9 series 14 - Hellfire 15 - Gemroad 16 - et al. antitank missile 17 - cruise missiles 18 - air-to-ground 19 - air defense missiles 20 - ground-to-air 21 - air-to-air 22 - inertial guidance + terrain-matching + infrared imaging terminal guidance 23 - infrared imaging guidance 24 - optical/television or radar tracking + radio command 25 - television guidance 26 - optical tracking + radio command 27 - infrared/ultraviolet bichromatic homing 28 - infrared homing 29 - infrared 30 -laser semiactive 31 - laser-guided bomb quidance 32 - laser-driven beam

In the Gulf War, infrared guidance missiles played an important role with gigantic intimidation power. Table 8 shows the photoelectronic guidance weapons that emerged during the Gulf War. By using the example of the Bullpup missile, in the period with the most intensive air raids, upwards of 100Bullpup missiles were launched each day [31].

The precision guidance capability of infrared guidance weapons is relatively high, with high hit rates. Coordinated with other weapons, the infrared guidance weapon exerted its function in the war.

Variations in the modern war environment spurred the development of infrared night vision systems. To gain the initiative on the battlefield, troops should have fighting capability either day or night.

In the Gulf War, infrared night vision equipment exhibited high functions. The AH-64 Apache armed helicopter, and the UH60A Blackhawk helicopters of the United States are equipped with thermal imaging reconnaissance systems and advanced night vision devices. Although the Iraqi Soviet-made Mi-24 attack craft were also equipped with FLIR [forward-looking infrared] night vision,

their performance was poorer than those on the American craft. An advanced night vision equipment, the cat's-eye night vision instrument for pilots of the U.S. Navy and the Marine Corps, costs more than 700,000U.S. dollars per set [32]. Infrared night vision equipment is extensively used in various electronic warfare aircraft, such as the F-117, the F4G, and the F-16 fighters, as well as the EA-6B and the EF-111A of the United States Air Force, and the M-1 tanks of the U.S. Army.

VI. Several Suggestions for Future Developments

1. Emphasize the development of guidance technology

The Gulf War was the largest scale high-tech war for weaponry since World War II. Precision guidance of weapons was one of the features of the war. Long-wave infrared imaging guidance technology and bichromatic guidance technology for precision guidance in the future have had further development. Research on the staring-type focal-plane array and the graphic processing will be further upgraded. As to the development directions of guidance, on the one hand, novel detection devices are under development, and, on the other hand, further research into hybrid guidance, has developed toward the hybrid guidance of integrating radar, television, lasers, and infrared multiple detection and tracking methods. With respect to tactical missiles in China, especially air defense missiles, this is also the direction that we should pursue.

2. Emphasize the development of infrared materials production techniques.

To apply the advanced infrared focal-plane array technique, costs should be cut, and large-lot production can be carried out. For example, large-lot production of focal-plane arrays has been the emphasis in the United States, with research of key

techniques such as materials growth, and applying chemical vapor deposition and growth techniques of metal and organic compounds for use in fabricating HgCdTe epitaxial growth layers. Moreover, photoelectric diodes will be grown on substrates of buffer gallium arsenide. Moreover, some low-cost materials (such as platinum silicide and gallium arsenide) are to be researched and produced in missile guidance heads in order to cut costs. In the case of China, the HgCdTe production technology still lags behind the developed nations to some extent. While pushing the development of advanced production technology, importation of advanced technology for production in China is also a right approach, thus speeding up the upgrading of precision guidance performance of tactical weapons in China.

3. Infrared detection and tracking requires that data processing should keep pace in its advance.

VHSIC enables the concentrated type high-speed graphics processing. In the case of China, while developing the staring-type infrared array devices, research on other related supplementary techniques should also proceed, laying a foundation for artificial intelligence in missiles.

4. Development of infrared countermeasures technology is a factor that should not be overlooked.

The emergence of precision guidance munitions will surely give rise to countermeasures. The infrared guidance system promotes the development of infrared countermeasures technology. In the Gulf War, the infrared countermeasures techniques also were quite active. Wars in the future will involve comprehensive countermeasures of multiple sensors in the sea, land, and air, in coping with communications, radar, and the command and control network. We should carry out automatic search, acquisition, analysis, and discrimination of threats from multiple targets

from all directions and in all frequency bands. In our case, while developing precision guidance munitions, infrared countermeasures technology is also a factor that cannot be overlooked. We should conduct system verification and simulation research on infrared countermeasures.

5. Development of infrared night vision equipment should be placed on the agenda.

Night war will be the principal mode in wars of the future. By using the long-wave infrared detection techniques in night vision equipment, night warfare capability can be further upgraded. Advanced infrared night vision equipment was installed in various branches of the U.S. Armed Forces. In the case of China, mass production and deployment of night vision instruments will be included in the military program to upgrade the night warfare capability of the Chinese Armed Forces.

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